

In the Abstract:

Please replace the paragraph at page 16, lines 2 to 5, with a replacement paragraph amended as follows:

~~The invention relates to a gas turbine component, namely~~
A static gas turbine component, especially for an aircraft
~~engine, engine. According to the invention, the static gas~~
turbine component is formed at least partially region-wise
of metal foam. An abradable shroud lining of metal foam
and a carrier allow a radial through-flow of gas.

[RESPONSE CONTINUES ON NEXT PAGE]